

FINITE ELEMENT ANALYSIS OF SHEAR TRANSFER OFFIBROUS CONCRETE UNDER COMPRESSIVE STRESS

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ABSTRACT:

This study presents a numerical analysis of push-off and modified push-off fibrous concrete specimens. The investigated variables were, volume fraction of fibres, area of reinforcement, and the ratio of the compressive to the shear stress. Eight nodes isoparametric elements were used to represent concrete, and embedded bar elements to represent the reinforcement. The nonlinear behaviour of fibrous concrete in compression, tension and the reduction of the shear modulus due to cracking were taken into account.

The numerical results showed good agreement with the experimental results, the average ratio of the calculated to the experimental first crack shear stress is (0.955), and the average ratio of the calculated to the experimental shear strength was (0.981). The cracks patterns showed that the fibrous specimens have less numbe

rs of cracks than that of plain specimens at a certain shear force, and the presence of fibres delay the initiation the cracks. The calculated slip is slightly more than the experimental slip at early stage of loading and the difference increased prior to the failure load. Key Words: cement, contour, cracks' pattern, fibres, finite element, shear transfer.

التحليل بطريقة العناصر المحددة لانتقال القص للخرسانة الليفية

تحت تأثير اجهادات انضغاطيارجعفرالسليفاني

تقدم هذه الدراسة تحليلا عدديا لنماذج الدفع ونماذج الدفع المحورة للخرسانة الليفية. المتغيرات التي تم دراستها هي النسبة الحجمية للالياف، مساحة التسليح، نسبة اجهاد الانضغاط الى اجهاد القص. تم استخدام عناصر ذات ثمانية عقد لتمثيل الخرسانة وعناصر قضبان مطمورة لحديد التسليح. تم اخذ السلوك غير الخطي للخرسانة الليفية في النضغاط والشد وكذلك نقصان معامل القص بسبب التشقق نتائج التحليل العددي اظهرت توافقا جيد مع النتائج العملية، معدل نسبة النتائج العددية الى العملية لاجهادات القص في مرحلة التشقق تساوي (0.955) وللمقاومة العظمى تساوي (0.981). الشقوق في نماذج الخرسانة الليفية تكون أقل عددا من الشقوق في الخرسانة الاعتيادية في قوة قص معينة ووجود الالياف يؤخر ظهور الشقوق. الانزلاق المحسوب كانت اعلى قليلا من النتائج العملية في المراحل المبكرة من تسليط الاحمال وزاد الفرق بزيادة الاحمال قبل الفشل

INTRODUCTION:

A shear force which is transmitted across a specific shear plane is denoted as shear transfer. Examples of such situations are precast concrete connection, brackets, corbels, members with shear span less than the effective depth where pure or direct shear is more likely to occur, column footing connections subjected to high shear forces and concrete cast at different ages, [1-3]. In some cases a crack exists in the shear plane before any shear force is applied due to either temperature deformation or due to the existence of tension forces caused by shrinkage restraint. There are two mechanisms in which the shear transfer strength may develop

For specimens with shear reinforcement normal to the shear plane with no cracks along the shear plane, the governing role in strength development is the truss action, and in case where there is a crack along the shear plane before the application of load, for specimens with reinforcement normal to the shear plane, a shear friction mechanism can be used [4].

Steel fibers enhance the tensile, flexural and shear strengths of concrete besides increasing the strain capacity under all states of stresses [5]. The finite element method was used for the nonlinear analysis of fibrous concrete members, such as shallow beams, deep beam, beam-column connections, corbels, columns and slab [6,7] .

Muhoder and Meshri [7] used the finite-element method for the nonlinear analysis of push-off reinforced concrete specimens subjected to direct shear with or without lateral compressive stress. Nine-noded , two dimensional isoparametric elements were used to represent concrete and steel. The Mohr-coulomb criterion was used as a failure envelope: $\tau =$

$$C - S_n . \tan f \quad (1)$$

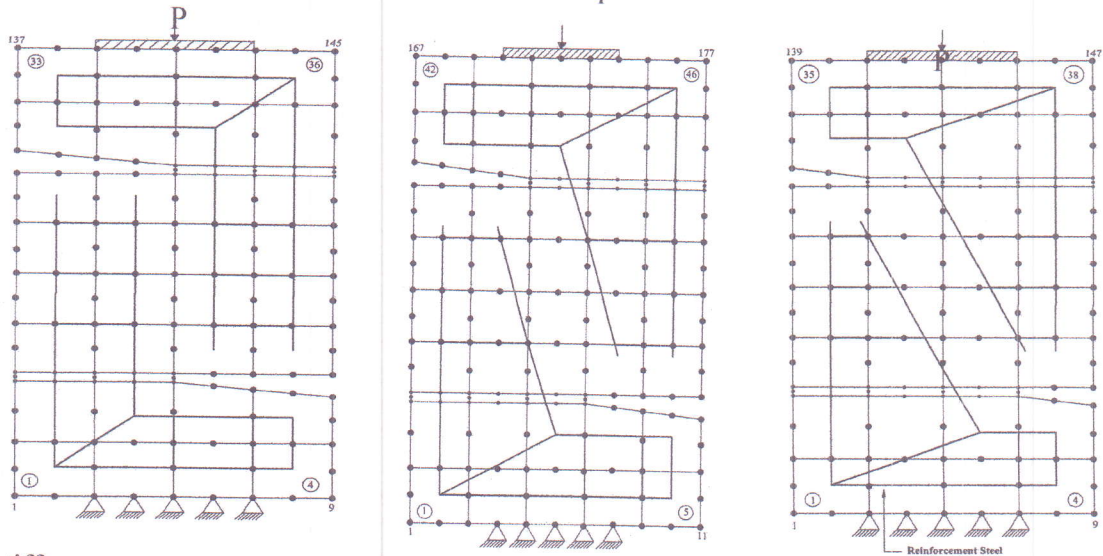
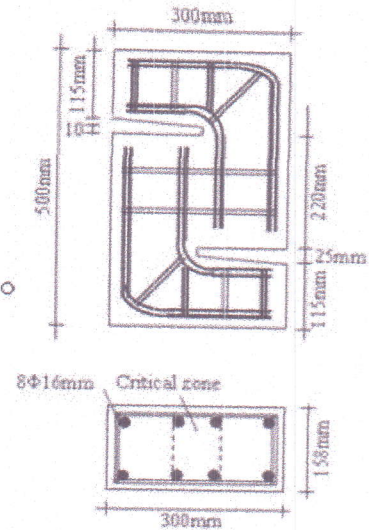
where τ = shear stress, S_n = normal stress (tension is positive), C = cohesion, f = angle of internal friction and $\tan f$ taken equal to 1.4 (normal weight concrete) as recommended by the ACI Code [1]. The ratio of the test / numerical shear strength for 15 specimens ranged between 0.94 to 1.3 with a standard deviation of 0.102. The numerical results showed that the shear stress distribution across the shear plane was intermediate between the uniform and parabolic distribution.

In this study, the main purpose is to investigate the influence of the steel fibers, reinforcement ratio, and the direct stresses on the shear transfer strength of push-off specimens by using the finite element method.

FINITE ELEMENT DESCRIPTION:

A nonlinear finite element analysis was carried out; the specimens were considered in a state of plane stress. Eight nodes isoparametric elements used to represent concrete, and embedded bar elements to represent the reinforcement.

The reinforcing bars are assumed as an axial member embedded anywhere within the element. This simulation was presented for the first time by Phillips and Zienkiewicz [8]. A perfect bond was assumed between concrete and the reinforcement. The modified Newton-Raphson algorithm is used in the present study, and the specimens



stiffness

Fig. (2a): Mesh representation Fig. (2b) Mesh representation Fig. (2c): Mesh representation of the push-off specimens of the push-off specimens of the push-off specimens

$$(q = 0)$$

$$(q = 15.25^\circ)$$

$$(q = 30.5^\circ)$$

matrix is modified in the second iteration of each load increment, a displacements convergence criterion is used.

The dimensions of the specimens is shown in Fig.(1), and Fig.(2a,b and c) show the mesh representation of the push-off specimens with three angles ($q = 0, 15.25^\circ$ and 30.5°). The figures show also the method of applying the loads on the specimens and the boundary conditions.

MATERIALS CONSTITUTIVE RELATIONSHIPS

The nonlinear behaviour of fibrous concrete in compression, tension and the reduction of the shear modulus due to cracking were taken into account. The materials constitutive relationships used including the uniaxial stress – strain relationships in compression and tension of the fibrous concrete and steel, biaxial behavior of the concrete, and failure criteria. **Compressive Stress-Strain Relationship:** Two models are used to represent the stress-strain curve; **A** – elastic – perfectly plastic, where the material is assumed elastic until the effective stress reach the compressive strength of fibrous concrete. **B** – strain –hardening approach where a parabolic stress – strain curve is assumed after the effective stress reach 30% of the compressive strength followed by a parabolic stress-strain curve and until the effective stress reach the compressive strength of fibrous concrete, then a perfect plasticity is assumed until crushing occurs.

In the absence of test data the empirical equations proposed by Soroushian and Lee [9] may be used to calculate the compressive strength of fibrous concrete as follow: $f_{cf} = f_c + 3.6V_f \frac{d_f l_f}{f}$ (2)

Where:

f_{cf} = compressive strength of fibrous concrete (MPa), f_c = compressive strength of plain concrete (MPa), V_f = fibers volume fraction, and l_f, d_f = length and diameter of the fibres respectively.

The strain at peak stress (ϵ_{cf}) as given by Soroushian and Lee [9] is used in this investigation:

$$\epsilon_{cf} = \frac{2E_f f_c}{E_c} + 0.0007V_f \frac{d_f l_f}{f} \quad (3)$$

The ultimate crushing strain in compression as defined in reference [5] is used:

$$\epsilon_{cuf} = 3011 + 2295V_f \text{ (micro strain)} \quad (4)$$

Modulus of Elasticity:

The following formula was used for the elastic modulus of plain concrete which depends on the concrete compressive strength and density [4].

$$E_c = \left(3320 \sqrt{f_c} + 6895 \right) (w_c / 2300)^{1.5} \quad (5)$$

w_c = concrete density in kg/m³

For fibrous concrete the modulus of elasticity can be calculated from equation [10]:

$$E_{cf} = \sqrt{E_{90}E_o} \quad (6)$$

where E_{cf} , E_{90} and E_o are secant moduli of elasticity of concrete with randomly oriented fibers, longitudinally aligned and transversely aligned fibers respectively.

$$E_o = \frac{E_c E_f}{V_c E_f + V_f E_c} \quad (7)$$

$$E_{90} = 2 \left[1 - \eta_f + \left(\eta_f \frac{M}{M_c} \right) \eta_c \right] \frac{E_c E_f}{2M_c + G_c} \quad (8)$$

where V_c and V_f are volume fractions of concrete and fibers.

G_c is concrete shear modulus.

η_c and η_f are the Poisson's ratio of concrete and fibres respectively.

$$M_c = 2(1 - \eta_c) \frac{E_c}{1 - \eta_c}, M_f = 2(1 - \eta_f) \frac{E_f}{1 - \eta_f} \quad (8)$$

Tensile Stress – Strain Relationship:

The stress – strain relationship of fibrous concrete in tension which is suggested in reference [11, 12], which can be represented by continuous function is used:

$$\frac{f}{f_{tf}} = \frac{b}{1 + e/(e_{tf})} \quad (9)$$

where $b = 1 + e/(e_{tf})$

$$b = 1.093 + 0.7132 R.I^{-0.926} \quad \text{For hooked fibres} \quad (10) \quad b = 1.093 +$$

$$7.4848 R.I^{-1.387} \quad \text{For smooth fibres} \quad (11) \quad b = 0.5811 + 1.093 R.I^{-}$$

$$0.7406 \quad \text{For crimped fibres} \quad (12)$$

$$R.I = W_f l_f / d_f$$

Where

$R.I$ is a reinforcing index and W_f is the weight fraction of the fibres. The tensile strain ϵ_{tf} at peak tensile stress f_{tf}' for fibrous concrete can be defined as proposed by Lee and Soroushian [9] are used in the present study as follow:

$$f_{tf}' = f_{tf}' \epsilon_c^{0.5} \left(1 + 0.016 N_f^{1.3} + 0.05 p d f_l f N_f^{0.5} \right) \quad (13) \quad \epsilon_{tf}' = \epsilon_t$$

$$\left(1 + 0.35 N_f d f_l f \right) \quad (14) \text{ where } f_{tf}' = \text{tensile strength of plain concrete. } \epsilon_t = \text{matrix cracking strain. } N_f = \text{Number of fibres per unit cross section area:}$$

$$N_f = h_0 \left(4 V_f / p d^2 f \right) \quad (15)$$

h_0 = Orientation factor, in the present study the value of h_0 used equal to 0.41 [13].

THE YIELD CRITERIA

Many yield criterions were proposed [14, 15] to define the plastic stage. In the present study the following yield function is used, which is adopted in many nonlinear analysis of concrete structures:

$$f(I_1, J_2) = [b_f(3J_2) + a_f I_1]^{0.5} = s_0 \quad (16)$$

where a_f and b_f are material parameters and s_0 is the uniaxial compressive strength. The material parameters have been obtained by Abdul- Razzak [5]:

$$1 - w^2$$

$$a_f = w^2 - 2w s_0 \quad (17)$$

$$b_f = w^2 - \frac{2w}{2w} \quad (18)$$

Where $w = e^x$.

The two dimensional stress representation [16] (biaxial compression-compression region) of fibrous concrete model is shown in Fig. (3). In the tension- compression zone, the tensile strength of fibrous concrete is predicted from a parabolic relation which was proposed by Al-Ta'an and Mahmood [17] as shown in Fig. (4) and is given by:

(19) $f_c f_{\mathfrak{d}}$ è

Equation (19) can be rewritten to give the peak compressive strength s_{2p} as a function of the uniaxial compressive strength f_{cf} , ratio of uniaxial compressive strength to its uniaxial tensile

$$s_{2p} = \frac{\sqrt{2(a_{22}s_2)}}{2} \quad (20)$$

and the peak tensile strength as:

$$s_{1p} = s_{2p} a_2$$

$$(21) \quad 4(a_2 - s_2)^{-1} \dot{u}$$

S_1P , $2P$ are the limiting principal tensile and compressive strength respectively.

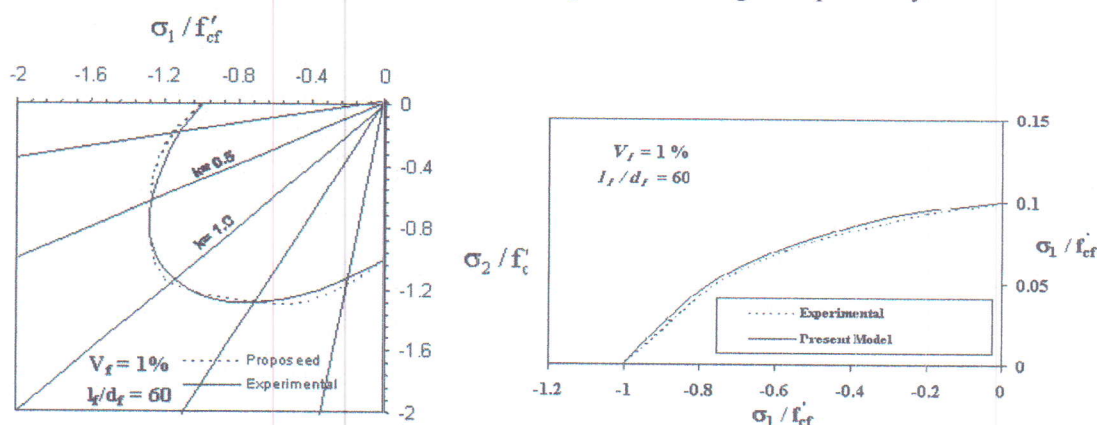


Fig. (3) Present model FRC for biaxial Fig. (4) Envelope for compression – compression. region biaxial tension – compression [17] [16]

RESULTS AND DISCUSSION

The properties of concrete used in tension and compression were taken from the experimental test results [18]. The cubes strength of plain and fibrous concrete was multiplied by 0.8 to obtain the equivalent cylinder strength, and the splitting strength also multiplied by 0.8 to get the direct tensile strength. Because the specimens were skew symmetry, it was necessary to analyze the entire specimen. The mesh used is shown in Fig. (2) for $\theta = 0, 15, 25^\circ$.

and 30.5°. Poisson's ratio was taken equal to 0.2. The steel properties used are shown in Table (1) for the longitudinal and transverse shear reinforcement [18].

Table (1): Properties of the Used Reinforcement

Diameter (mm)	Location	Yield strength (MPa)	Tensile strength (MPa)
16	Longitudinal	437	730
6	Stirrups	510	-----
10	Stirrups	321	438

1-Shear Transfer Strength

Table (2) shows comparison of the finite element results with the experimental results [18] of shear strength for the push-off specimens. The numerical results showed a very good agreement with the test results for all specimens. The ratio of (v_u cal.) to (v_u exp.) range from (0.89-1.08) with an average value of (0.981), standard deviation (SD) = 0.0485 and a coefficient of variation of 4.9%.

Table (2) Comparison of the Numerical and Experimental Results Of Shear Strength for Pushoff Specimens

Specimen No.	Shear plane length(mm)	No.and dia. of stirrups	Fibres volume %	F.E. Shear strength MPa	Experimental Shear strength MPa[18]	$v_{u\text{cal}}/v_{u\text{exp}}$
A1	220	0	0	4.83	4.95	0.98
A2	230	0	0	8.59	8.82	0.97
A3	255.5	0	0	13.57	13.02	1.04
A4	220	0	0.5	5.50	5.20	1.06
A5	230	0	0.5	9.25	9.32	0.99
A6	255.5	0	0.5	14.26	15.17	0.94
A7	220	0	1.0	6.49	6.55	0.99
A8	230	0	1.0	10.60	11.16	0.95
A9	255.5	0	1.0	15.11	15.38	0.98
A10	220	0	1.5	6.40	6.27	1.02
A11	230	0	1.5	10.60	11.23	0.94
A12	255.5	0	1.5	15.87	15.90	1.00
B1	220	3-6mm	0	6.44	6.63	0.97
B2	230	3-6mm	0	11.22	11.37	0.99
B3	255.5	3-6mm	0	14.70	13.61	1.08
B4	220	3-6mm	0.5	6.50	6.90	0.94
B5	230	3-6mm	0.5	11.88	11.75	1.01
B6	255.5	3-6mm	0.5	15.85	16.02	0.99
B7	220	3-6mm	1.0	7.50	8.04	0.93
B8	230	3-6mm	1.0	12.50	11.75	1.06

B9	255.5	3-6mm	1.0	16.84	17.13	0.98
B10	220	3-6mm	1.5	8.06	8.96	0.90
B11	230	3-6mm	1.5	12.50	11.75	1.06
B12	255.5	3-6mm	1.5	18.80	18.34	1.02
C1	220	2-10mm	0	8.85	8.86	1.00
C2	230	2-10mm	0	11.56	12.63	0.92
C3	255.5	2-10mm	0	16.10	16.02	1.00
C4	220	2-10mm	0.5	8.91	9.47	0.94
C5	230	2-10mm	0.5	12.33	12.90	0.96
C6	255.5	2-10mm	0.5	16.8	16.40	1.02
C7	220	2-10mm	1.0	8.90	9.60	0.93
C8	230	2-10mm	1.0	13.10	14.58	0.90
C9	255.5	2-10mm	1.0	18.90	18.40	1.03
C10	220	2-10mm	1.5	9.00	10.10	0.89
C11	230	2-10mm	1.5	13.87	14.72	0.94
C12	255.5	2-10mm	1.5	19.30	19.18	1.01

Fig. (5) shows comparison of the first crack shear stress of the experimental and numerical results for group A, with ($\rho_f = 0$) for fibres percentage of 0, 0.5, 1 and 1.5. The figure shows that the experimental results are in close agreement with the numerical results. The average of the calculated to the experimental first crack shear stress is 0.955.

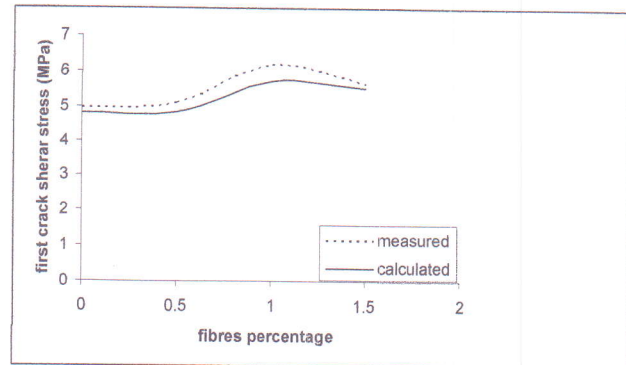


Fig. (5) Comparison of experimental and calculated first crack shear stress for Push-off specimens A1, 4, 7 and 10

2-Effect of Reinforcement Parameter
The effect of the reinforcement parameter ρ_f on the shear strength is shown in Table (2) for fibres percentages of (0, 0.5, 1, and 1.5). The results show that the shear strength increases with the reinforcement parameter for all fibres percentages.

3-Effect of Steel Fibres

Table (2) shows that the ultimate shear strength increased with the fibres volume increase in specimens with or without stirrups. The results showed that sometimes fibres was as effective as stirrups in increasing the shear strength, the same results obtained for all specimens.

4- Shear Stress Distribution

To show the distribution of the shear stress across the shear zone, contour of the shear

stresses are plotted for selected specimens to show the fibres effect on the shear stress distribution. Fig. (6) shows the shear stress contours and shear stress distribution across the shear plane of specimen A1 (without fibres). The figure shows that the shear stresses are concentrated near the slots (at the shear zone boundaries) and decreased toward the center of the shear plane.

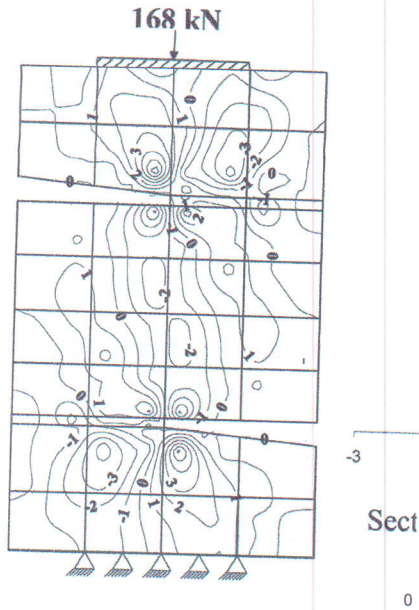


Fig. (6) Contour of the shear stresses of push – off
Specimen A1 $V_f = 0$, $q = 0$

Fig. (7) shows the shear stresses contour for specimen A7 ($V_f = 1\%$) for loads 192 kN and 224 kN respectively. The two figures have the same trend of the shear stress distribution, except that the shear stress is higher for higher load (224 kN). The figures show also the shear stress distribution along the shear plane .

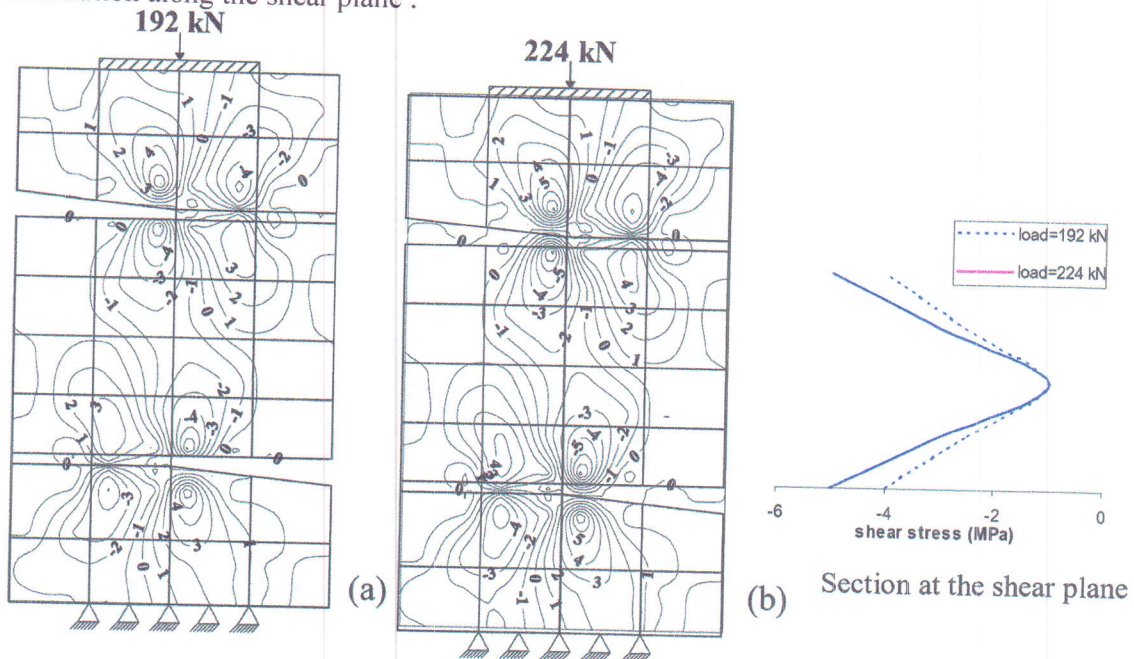


Fig. (7) Contour of shear stresses of push – off specimen A7
 $V_f = 1\%$, $q = 0$

Fig. (8) Shows the contour of the shear stresses for specimens A2 ($V_f = 0$ and $\Theta = 15.25^\circ$) the figure shows that the contour of the shear stress is the same as that of specimen A1 Fig. (6), but the shear stress along the shear plane is slightly different, this may be attributed to the lateral compressive stress. Fig. (9) shows the contour of the shear stresses for specimen A8 ($V_f = 1\%$),

the figure shows that the maximum shear stress increased near the slots and decreased toward the center of the shear plane.

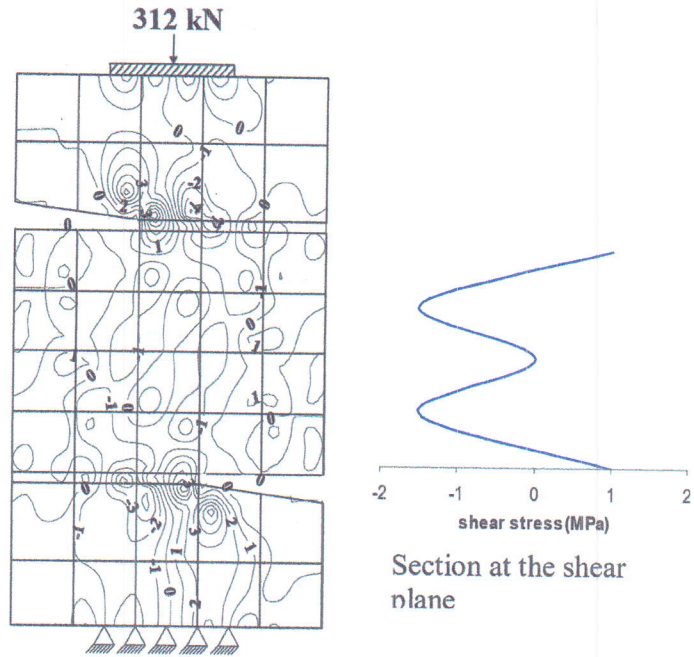


Fig. (8) Contour of the shear stresses of push – off specimen A2, $V_f = 0$, $q = 15.25^\circ$

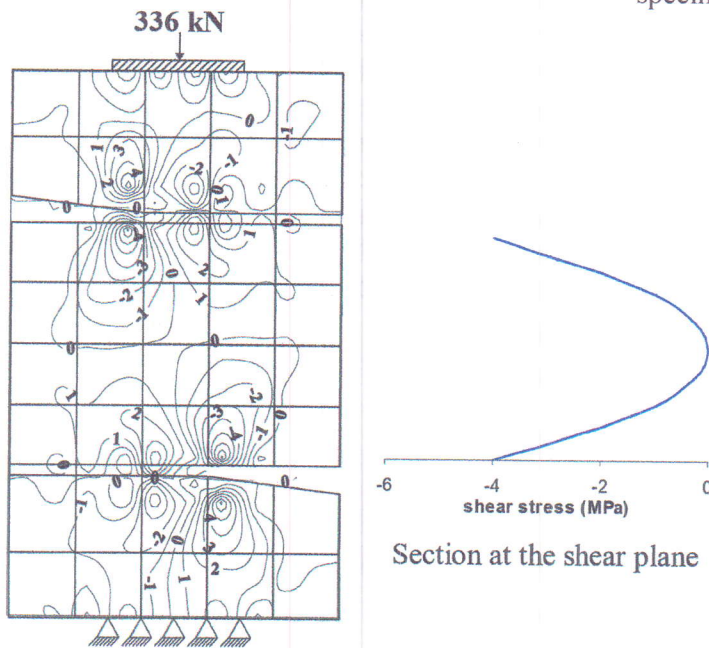


Fig. (9) Contour of the shear stress of push – off specimens A8, $V_f = 1\%$, $q = 15.25^\circ$

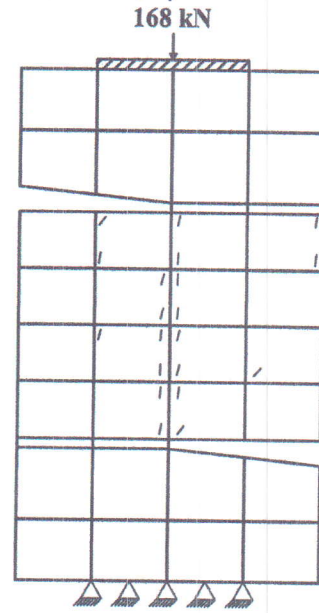


Fig. (10) Cracks pattern of specimen A1, $V_f = 0$

5- Cracks Patterns

The cracks were drawn at the mid position only (between the two slots) for selected specimens to show the fibres effect on the initiation and propagation of cracks.

specimen A2, $V_f=0$

(a) (b)
Fig.(13): Cracks pattern of specimen A11, $V_f=1.5$

6- Slip

The slips of selected specimens were presented below. Fig.(14) shows the slip of specimen A10 ($V_f=1.5$). The figure shows that the calculated slip is close to the experimental slip at early stage and slightly differ at failure, this variation may be attributed to the representation of the shear reinforcement in the finite element method which is represented by a line element with axial stiffness only and differ from the actual shear reinforcement. Fig.(15) shows the slip of specimen C10, the figure shows that the calculated slip is slightly differ from the experimental slip, this may be due to the same reason as for specimen A10, that is the representation of the shear reinforcement.

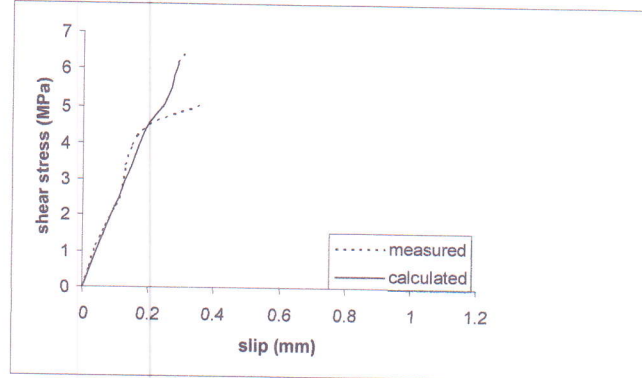


Fig. (14): Shear stress- slip for specimen A10($V_f=1.5\%$)

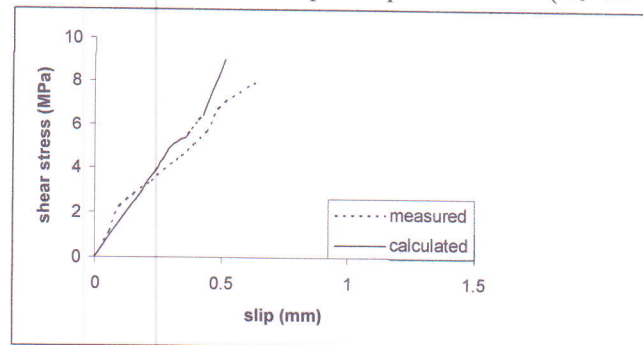


Fig.(15) shear stress-slip for specimen C10 ($V_f=1.5\%$)

CONCLUSIONS

- 1- The finite element results of the first cracking and the ultimate shear stress showed good agreement with the experimental results, the average ratio of the calculated shear strength to the experimental shear strength was (0.9814).
- 2- The distribution of the shear stress of the push-off specimens is maximum at the ends of the shear plane and decreased towards the centre.
- 3- The predicted slip of specimens without shear reinforcement push-off results showed good agreement with the experimental slip, while the slip of specimens with shear reinforcement slightly differs from the experimental slip.
4. The cracks initiation and propagation pattern showed a realistic sequence.

5. The nonlinear finite element analysis can be used to trace the behaviour of shear transferspecimens and can be used for a parametric study.

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